Joint Motion

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Introduction

 This presentation addresses the kinematic equations for n-body systems whose joints can be any of the following types

❖ 3 dof rotational ← joint type(JT)=4

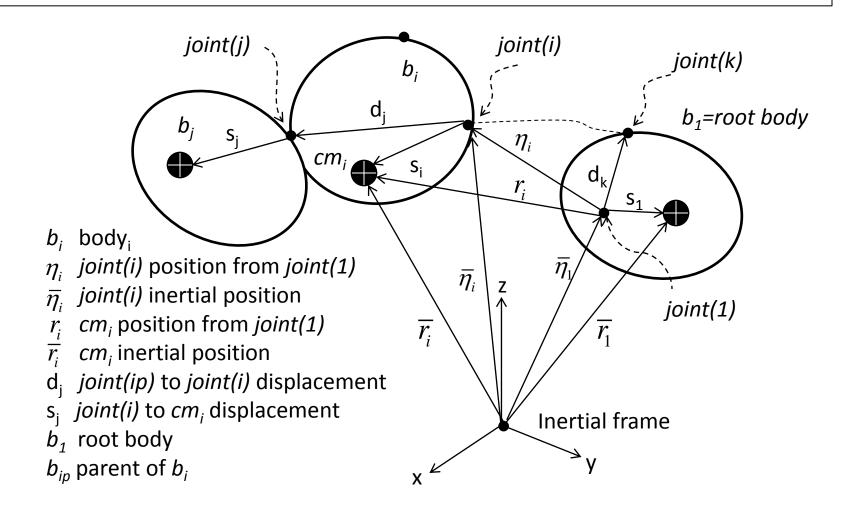
 MBS with Mixed joint types appear often in robotics, and complex vibration isolation mechanisms. The generalized coordinates chosen for the considered system is the set:

$$q = \left\{ \gamma_i \right\}_{i=1:N}$$

where

$$\gamma_{i} = \begin{cases} \delta_{i}, \text{ scalar displacement;} & \text{JT}(i) = 1\\ \mathbf{x}_{i}^{i}, 3 \times 1 \text{ displacement;} & \text{JT}(i) = 2\\ \theta_{i}, \text{ joint angle;} & \text{JT}(i) = 3\\ \tilde{\varepsilon}_{i}, \text{ relative attitude quaternion;} & \text{JT}(i) = 4\\ = \text{inboard joint coordinate of } b_{i} \end{cases}$$

Fig. 1 Joint 1 Centric Notation



- The root body b_1 is the reference body whose position and attitude serve as the starting value to compute the same for other bodies in the system in a hierarchical manner. Generally, the choice of b_1 is arbitrary. In the case of a humanoid robot, b_1 could be the head, the torso or the hip.
- Body indexing rule used here is the Parent-First order meaning that the index of a body is always a lower integer number than the indices of its children.
- The chain of bodies between b_1 and b_j shall be denoted as $\{i \mid i \leq j\}$ or just $i \leq j$. The less-than-or-equal relation over body indices is a topological order and not a numerical order.
- The set of bodies branching from b_j shall be denoted as $\{i \mid i \geq j\}$ or just $i \geq j$. The greater-than-or-equal relation over body indices is a topological order and not a numerical order.
- All vectors in the following discussion are given in the format x_j^i . The subsript j denotes the body that x belongs to and the superscript i denotes the coordinate frame that the vector is in.
- Vectors with no superscript are given in inertial coordinates unless defined otherwise

Fig. 2 Prismatic Joint, JT=1

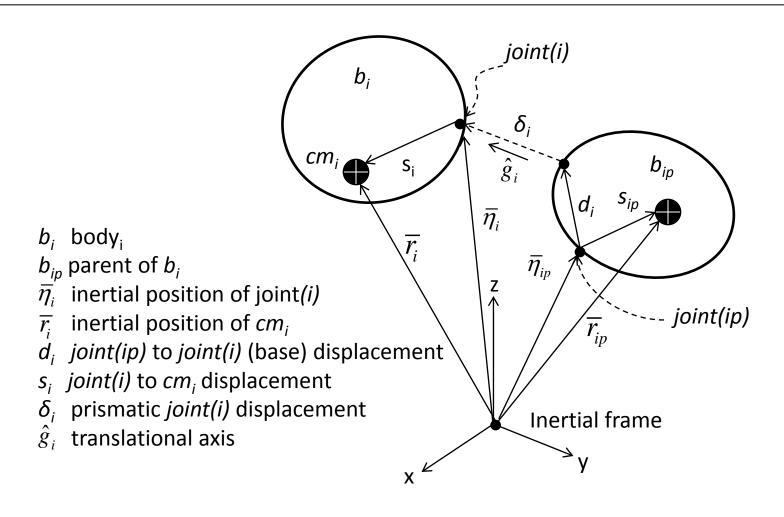


Fig. 3 3 Dof Translational Joint, JT=2

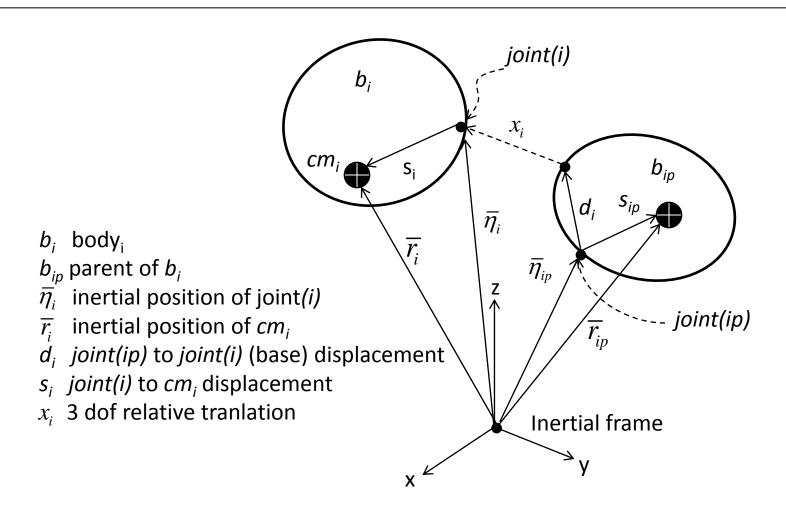


Fig. 4 1 Dof Rotatioal Joint, JT=3

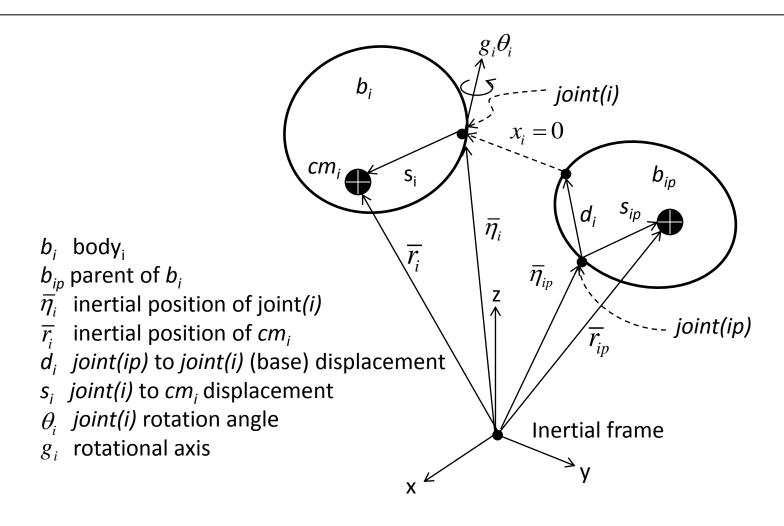
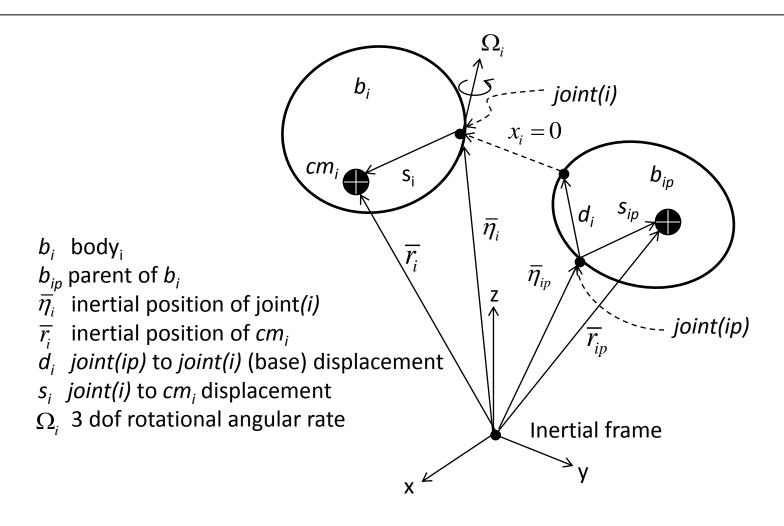


Fig. 5 3 Dof Rotatioal Joint, JT=4



Coordinate Frames

- A general reference frame is a 3 dimensional space with its origin defined at some point of interest. The translational motion of objects are defined by the [x, y, z] coordinates of the objects over time in that frame. The attitude motion of each body is defined by a direction cosine matrix that maps vectors fixed on that body to the reference frame over time.
- A local reference frame (LRF) shall mean a body fixed reference frame whose origin is located at the inboard joint of the body.
- An inertial frame is any frame in which Newtonian dynamics hold.
- Workspace frame is an inertial reference frame chosen to define the motion of a mechanism or an object such as a robot in that frame.

Attitude Equations

- A local reference frame is a body fixed three dimensional coordinate frame with its origin at the inboard joint of the body. For b₁, the LRF₁ origin is defined at an arbitrary reference point on it, called joint(1) here. See Fig. 1.
- Given $q = \{\gamma_i\}_{i=1:N}$ the direction cosine matrix C_i from b_i to the inertial frame is defined recursively by

$$C_i = C_{ip}C_i^{ip}(\gamma_i) \text{ for } i \ge 2$$
 (2)

where $C_i^{ip}(\gamma_i)$ = relative dcm from b_i to b_{ip} frame

 $ip = index of parent body of b_i$

 $C_1 = C_1^0$, 0 superscript means the inertial frame

• If joint(i) is translational (JT(i)=1 and 2)

$$C_i = C_{ip} \text{ and } C_i^{ip}(\gamma_i) = e_{3\times 3}$$
 (2a)

• If joint(i) is 1 dof rotational with $\gamma_i = \theta_i$ (JT(i)=3), then

$$C_i^{ip}(\theta_i) = \overline{C}_i^{ip} \left[(1 - \cos(\theta_i)) g_i^i g_i^{iT} + \cos(\theta_i) e + \sin(\theta_i) \tilde{g}_i^i \right]$$
 (2b)

where $\bar{C}_i^{ip} = C_i^{ip} (\theta_i = 0)$

 $g_i^i = \text{joint}(i)$ rotation axis in LRF_i coordinates

 \tilde{g}_{i}^{i} = skew symmetric matrix of vector g_{i}^{i}

 $e = 3 \times 3$ identity matrix

If joint(i) is 3 dof rotational with the relative quaternion $\gamma_i = \tilde{\varepsilon}_i = (a \ b \ c \ d)^T$, (JT(i)=4) then

$$C_{i}^{ip}(\tilde{\varepsilon}_{i}) = \begin{bmatrix} a^{2} - b^{2} - c^{2} + d^{2} & 2(ab - cd) & 2(ac + bd) \\ 2(ab + cd) & -a^{2} + b^{2} - c^{2} + d^{2} & 2(bc - ad) \\ 2(ac - bd) & 2(bc + ad) & -a^{2} - b^{2} + c^{2} + d^{2} \end{bmatrix}$$
(2c)

• Given $q = \{\gamma_i\}_{i=1:N}$, all $\{C_i(\gamma_i)\}_{i=1:n}$ can be computed by Eqs. (2,2a,2b,2c).

Position Equations

- Given $\{C_i(\gamma_i)\}_{i=1:n}$, all body fixed vectors in the system can be expressed in inertial coordinates
- All joint and cm positions $\{\overline{\eta}_i, \overline{r}_i\}_{i=1:n}$ in the inertial reference frame per Figs. 1 and 2 can be computed as

$$\overline{\eta}_{i} = \begin{cases}
\overline{\eta}_{ip} + d_{i} + \hat{g}_{i} \delta_{i} &, \text{ if } \gamma_{i} = \delta_{i} \\
\overline{\eta}_{ip} + d_{i} + C_{ip} x_{i}^{i} &, \text{ if } \gamma_{i} = x_{i}^{i} \\
\overline{\eta}_{ip} + d_{i} &, \text{ if } \gamma_{i} = \theta_{i} \text{ or } \overline{q}_{i}
\end{cases}$$
(3)

for i = 2: N with $\overline{\eta}_{1p} = 0$

$$\overline{r_i} = \overline{\eta_i} + s_i$$
, for $i = 1:N$ (4)

where $d_i = \text{nominal } \overline{\eta}_i \text{ position from } \overline{\eta}_{ip} \text{ when } x_i^i = 0 \text{ or } \delta_i = 0$;

 \hat{g}_i = slider axis; δ_i = linear displacement

 $s_i = cm_i$ position from $\overline{\eta}_i$; $x_i^i =$ displacement vector in b_{ip} coordinates

 $\overline{\eta}_i = \text{inertial position of joint}(i)$

 \overline{r}_i = inertial position of cm_i

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Velocity Equations

• The system rates state $\dot{q}^* = \left\{\dot{\gamma}_i^*\right\}_{i=1:N}$ is either obtained by solving the equations of motion or prescribed, with

$$\dot{\gamma}_{i}^{*} = \begin{cases} \dot{\delta}_{i} & \text{if } \gamma_{i} = \delta_{i} (=> \Omega_{i}^{i} = 0) \text{ ; JT=1} \\ \dot{x}_{i}^{i} & \text{if } \gamma_{i} = x_{i}^{i} (=> \Omega_{i}^{i} = 0) \text{ ; JT=2} \\ \dot{\theta}_{i} & \text{if } \gamma_{i} = \theta_{i} \text{ (}=> \dot{x}_{i}^{i} = 0) \text{ ; JT=3} \\ \Omega_{i}^{i} & \text{if } \gamma_{i} = \overline{q}_{i} \text{ (}=> \dot{x}_{i}^{i} = 0) \text{ ; JT=4} \end{cases}$$

 Ω_i^i = angular rate of LRF_i w.r.t LRF_{ip} in LRF_i coordinates

• $\dot{\tilde{\varepsilon}}_i$ relates to Ω_i^i for JT=4 as

$$\dot{\tilde{\varepsilon}}_i = \frac{1}{2} E_i \Omega_i^i \tag{5}$$

where
$$\tilde{\varepsilon}_i = [a, b, c, d]^T$$
 and $E_i = \begin{bmatrix} d & -c & b \\ c & d & -a \\ -b & a & d \\ -a & -b & -c \end{bmatrix}$

• Given $\dot{q}^* = \left\{\dot{\gamma}_i^*\right\}_{i=1:N}$ all body angular velocities can be computed by

$$\omega_{i} = \begin{cases} \omega_{ip} & \text{JT}(i)=1 \text{ or } 2\\ \omega_{ip} + g_{i}\dot{\theta}_{i} & \text{JT}(i)=3\\ \omega_{ip} + \Omega_{i} & \text{JT}(i)=4 \end{cases}$$
(6)

for i = 2: N, with $\omega_{1p} = 0$

where $g_i = \text{inboard 1 dof joint rotational axis of } b_i$

$$\Omega_i = C_i \Omega_i^i$$
; relative angular rate in inertial coordinates (7)

• Given $\{\dot{\bar{\eta}}_1,\omega_i\}_{i=1:n}$ the inertial velocities of all joints can be computed per Figs. 1 and 2 as

$$\dot{\overline{\eta}}_{i} = \begin{cases}
\dot{\overline{\eta}}_{ip} + \omega_{ip} \times \overline{d}_{i} + \hat{g}_{i} \dot{\delta}_{i} & JT(i) = 1 \\
\dot{\overline{\eta}}_{ip} + \omega_{ip} \times \overline{d}_{i} + C_{ip} \dot{x}_{i}^{i} & JT(i) = 2 \\
\dot{\overline{\eta}}_{ip} + \omega_{ip} \times \overline{d}_{i} & JT(i) = 3 \text{ or } 4
\end{cases}$$
(8)

for i = 2: N with $\dot{\overline{\eta}}_{1p} = 0$

where

$$\overline{d}_{i} = \begin{cases}
C_{ip}(d_{i}^{i} + \hat{g}_{i}^{i}\delta_{i}) & \text{JT}(i)=1 \\
C_{ip}(d_{i}^{i} + x_{i}^{i}) & \text{JT}(i)=2 \\
C_{ip}d_{i}^{i} & \text{JT}(i)=3 \text{ or } 4
\end{cases}$$

• Given $\{\dot{\bar{\eta}}_i\,,\,\omega_i\}_{i=1:n}$ the inertial velocities of all body cm's can be computed follows

$$\dot{\overline{r}}_i = \dot{\overline{\eta}}_i + \omega_i \times s_i \quad , \text{ for } i = 1:N$$

Acceleration Equations

• Given $\{\dot{\omega}_1, \ddot{\gamma}_i^*\}_{i=1:N}$ all angular accelerations can be computed recursively as

$$\dot{\omega}_{i} = \begin{cases} \dot{\omega}_{ip} & \text{JT}(i)=1 \text{ or } 2\\ \dot{\omega}_{ip} + g_{i} \ddot{\theta}_{i} + \omega_{ip} \times g_{i} \dot{\theta}_{i} & \text{JT}(i)=3\\ \dot{\omega}_{ip} + C_{i} \dot{\Omega}_{i}^{i} + \omega_{ip} \times C_{i} \Omega_{i}^{i} & \text{JT}(i)=4 \end{cases}$$

$$\text{for } i = 2: N$$

$$(10)$$

• Given $\{\ddot{\overline{\eta}}_1,\dot{\omega}_i\}_{i=1:N}$ and Eqs. (7, 8,9) all joint accelerations can be computed recursively as

$$\ddot{\overline{\eta}}_{i} = \begin{cases}
\ddot{\overline{\eta}}_{ip} + \dot{\omega}_{ip} \times \overline{d}_{i} + \hat{g}_{i} \ddot{\delta}_{i} + \omega_{ip} \times (\omega_{ip} \times \overline{d}_{i}) + 2\omega_{ip} \times \hat{g}_{i} \dot{\delta}_{i} \Leftrightarrow (JT(i)=1) \\
\ddot{\overline{\eta}}_{ip} + \dot{\omega}_{ip} \times \overline{d}_{i} + C_{i} \ddot{x}_{i}^{i} + \omega_{ip} \times (\omega_{ip} \times \overline{d}_{i}) + 2\omega_{ip} \times \dot{x}_{i} \Leftrightarrow (JT(i)=2) \\
\ddot{\overline{\eta}}_{ip} + \dot{\omega}_{ip} \times \overline{d}_{i} + \omega_{ip} \times (\omega_{ip} \times \overline{d}_{i}) & \Leftrightarrow (JT(i)=3 \text{ or } 4)
\end{cases}$$

$$for i = 2:n$$

• Given $\{\dot{\bar{\eta}}_i, \omega_i\}_{i=1:n}$, the cm inertial accelerations can be computed as

$$\ddot{r}_{i} = \ddot{\eta}_{i} + \dot{\omega}_{i} \times s_{i} + \omega_{i} \times (\omega_{i} \times s_{i}) \quad \text{for } i = 1:n$$
(12)

Other Points of Interest

- Motion of position markers on bodies may be needed for performance or constraint evaluations.
- Calculating the k-th position marker on b_i for some j:

$$p_{i}(k) = C_{i} p_{i}^{j}(k) \tag{13}$$

$$\overline{p}_{i}(k) = \overline{\eta}_{i} + p_{i}(k) \tag{14}$$

$$\dot{\bar{p}}_{j}(k) = \dot{\bar{\eta}}_{j} + \omega_{j} \times p_{j}(k) \tag{15}$$

$$\ddot{\overline{p}}_{j}(k) = \ddot{\overline{\eta}}_{j} + \dot{\omega}_{j} \times p_{j}(k) + \omega_{j} \times (\omega_{j} \times p_{j}(k))$$
(16)

where $p_{j}^{j}(k) = \text{position of marker}(k) \text{ from joint}(j) \text{ in } b_{j} \text{ coordinates}$

$$p_i(k)$$
 = inertial coordinates of $p_i^j(k)$

$$\overline{p}_i(k)$$
 = inertial position of position marker(k)

$$\dot{\bar{p}}_{i}(k)$$
 = inertial velocity of position marker(k)

$$\ddot{p}_{i}(k)$$
 = inertial acceleration of position marker(k)

- Motion of directional markers on member bodies may also be needed for performance or in constraint evaluations.
- Caculating the k-th directional marker on b_i for some j:

$$u_{i}(k) = C_{i}u_{i}^{j}(k) \tag{17}$$

$$\dot{u}_{i}(k) = \omega_{i} \times u_{i}(k) \tag{18}$$

$$\ddot{u}_{i}(k) = \dot{\omega}_{i} \times u_{i}(k) + \omega_{i} \times (\omega_{i} \times u_{i}(k))$$
(19)

where $u_j^j(k)$ = unit vector of direction marker(k) on b_j in b_j coordinate

 $u_j(k)$ = inertial coordinates of $u_j^j(k)$

 $\dot{u}_{i}(k)$ = inertial velocity (turning rate) of $u_{i}(k)$

 $\ddot{u}_j(k)$ = inertial acceleration of $u_j(k)$

Examples

- A ball joint is one with the coordinate $\gamma_i = \overline{q}_i$.
- A slider joint is one with the coordinate $\gamma_i = \delta_i \in \mathbb{R}$
- A universal joint consists of two serially linked joints of b_{ip} and b_i such that

$$(\gamma_{jp}, \gamma_j) = (\theta_{jp}, \theta_j)$$
 with $g_j^T g_{jp} = 0$ and $d_j = 0_{3\times 1}$

• A 6 dof joint consists of two serial joints of b_{jp} and b_j such that

$$(\gamma_{jp}, \gamma_j) = (\mathbf{x}_{jp}^{jp}, \overline{q}_j), \text{ with } d_j = 0$$

Summary

- The kinematics equations in attitude, position, velocity and acceleration have been presented for a rigid joint-connected multibody system that has different types of joints. These joints can have either 1 or 3 dofs and are either rotational or translational. The kinematic equations are defined recursively and conditionally based on the type of joint.
- Kinematics equations for the position and directional markers have been presented for the purpose of performance and constraint evaluations.